



Improved Materials and Processes for Wind Turbine Blades

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JEC Wind Energy Forum

Agenda

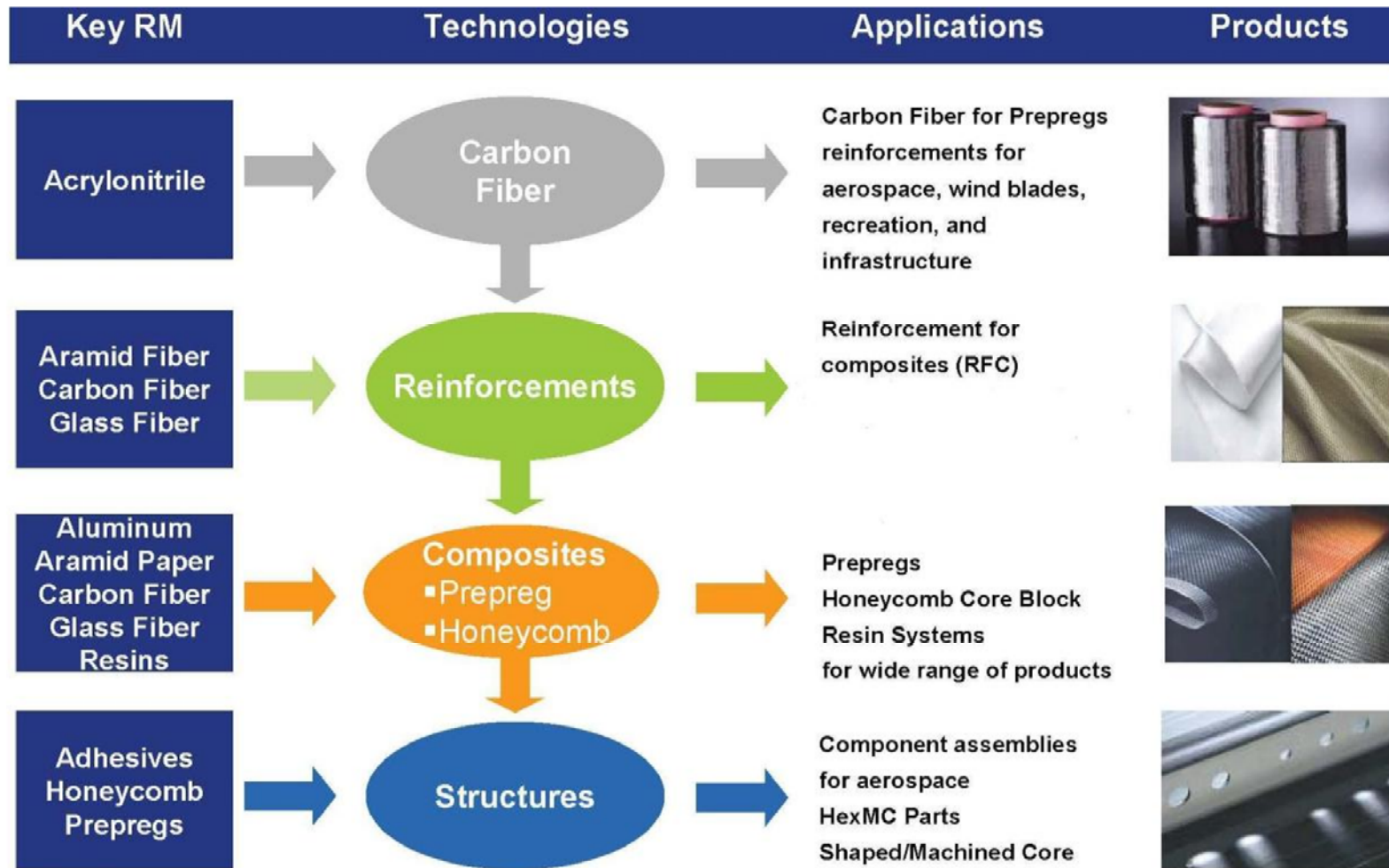
- **Introduction to Hexcel**
- **Prepreg technology**
- **Prepregs for the surface**
- **Prepregs for the load-carrying structure**
 - Comparisons between infusion and prepreg technologies
- **Use of prepregs for construction of spar caps**
 - Conventional, pre-cured
 - New: co-infused and co-cured
- **Conclusions**

Company Profile

- **Leading global provider of advanced composites**
- **Technology leader with largest portfolio of qualifications**
- **Primary markets: commercial aerospace, space & defense and industrial**
- **Net Sales of \$1,392.4 million in 2011**
- **Approximately 4,000 employees worldwide**
- **18 production sites (including JV in Malaysia)**
- **Headquarters in Stamford, CT, USA**
- **Listed on NYSE**



Hexcel - Vertically Integrated



Hexcel in Global Wind Energy

- **Market Leader for prepreg materials in Wind Energy**
- **Annual capacity of >20 000t**
- **Over 20 years experience**
- **Global Supply, Sales, Technical Support and R&T**
- **Product development in close cooperation with key accounts**



Plant for wind energy at Windsor
Colorado, opened in 2009
(Other dedicated plants in Austria
and in Tianjin, China)

Prepreg Technology

What are 'prepregs'?

Impregnation of Fibre and Fabrics with Resin



**Prepreg production is
now highly industrialised
for optimum cost and quality**

Typical Prepreg Systems in Wind Energy

Resin systems

M9G 310 J/g

M9GF 230 J/g

M19G 160 J/g

UD Products

Carbon 500-600 gsm

Glass 1000-3000 gsm

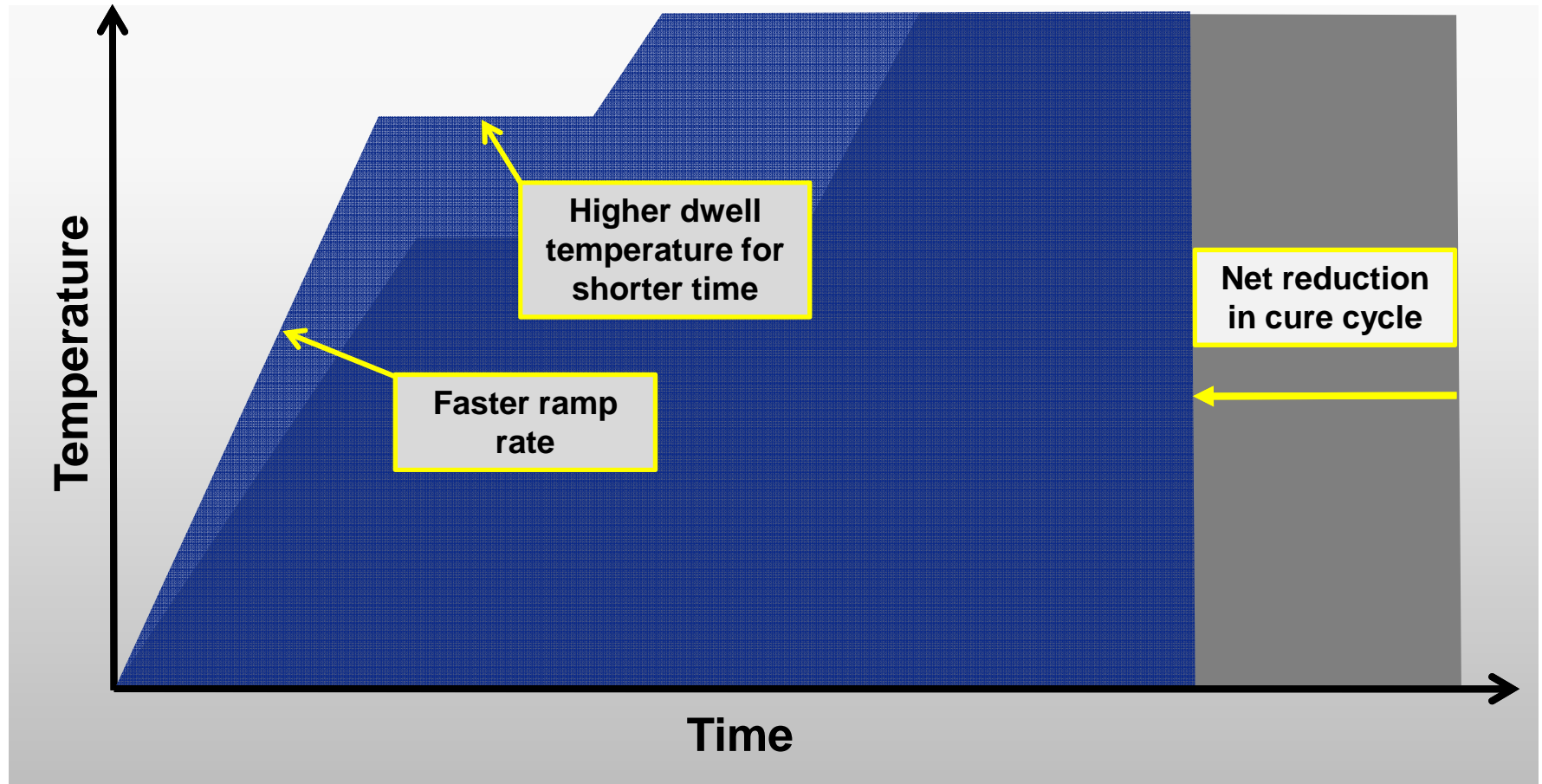
Overall cure cycles

~4 to ~8 hours (optimisation is key)

Storage	
Temperature	Shelf life
+23°C	6 weeks
+5°C	6 months
-18°C	18 months

**Typical prepregs:
high areal weights + full impregnation + low reaction enthalpy**

The Value of Low Exotherm in Thick Parts

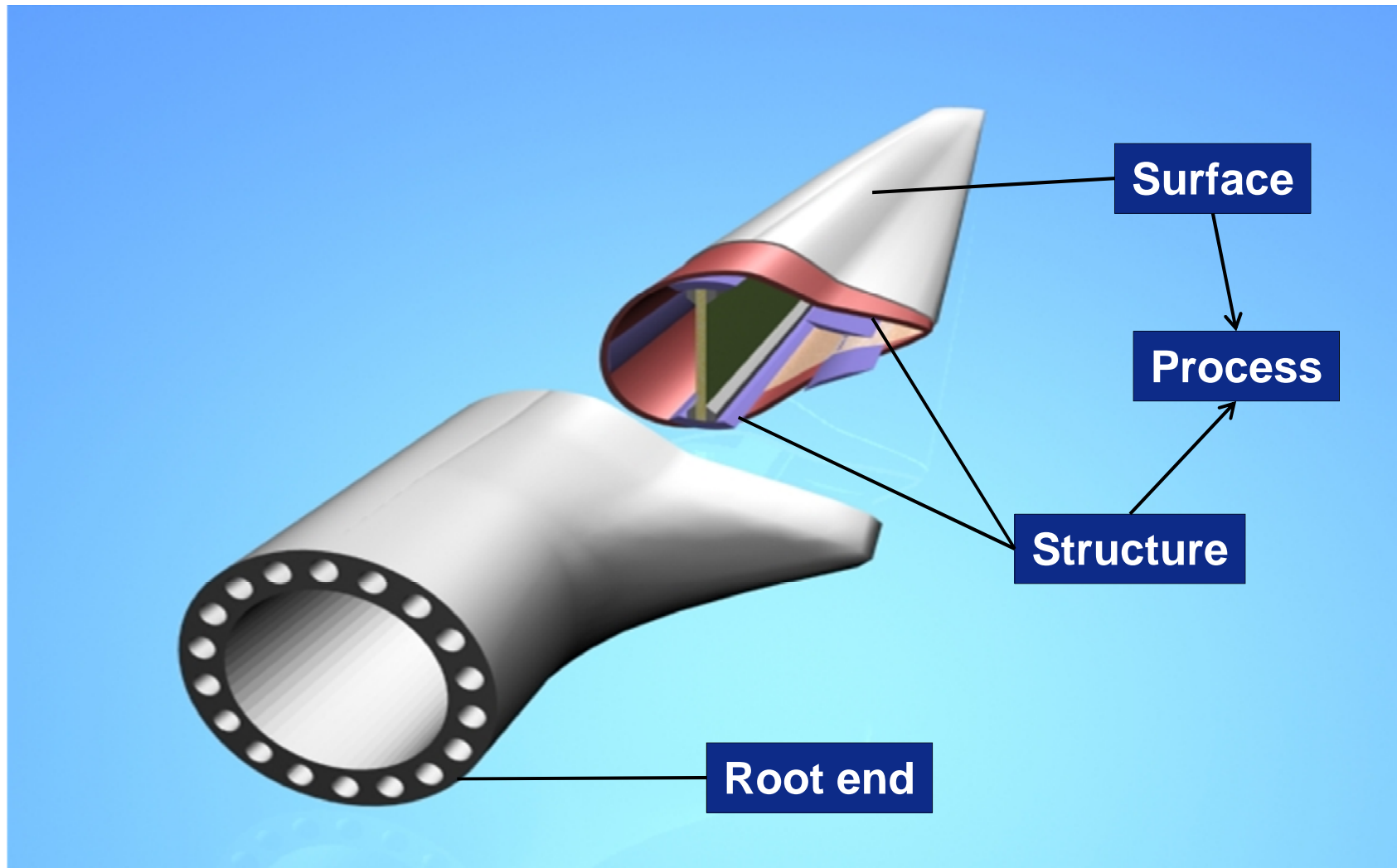


Low exotherm matrix e.g. M19G



Standard exotherm matrix e.g. M9G

Major Features of Typical Wind Turbine Blades



Prepregs for the Surface

Prepregs for the Shell Surface

Shell prepregs are used for the aerodynamic shell

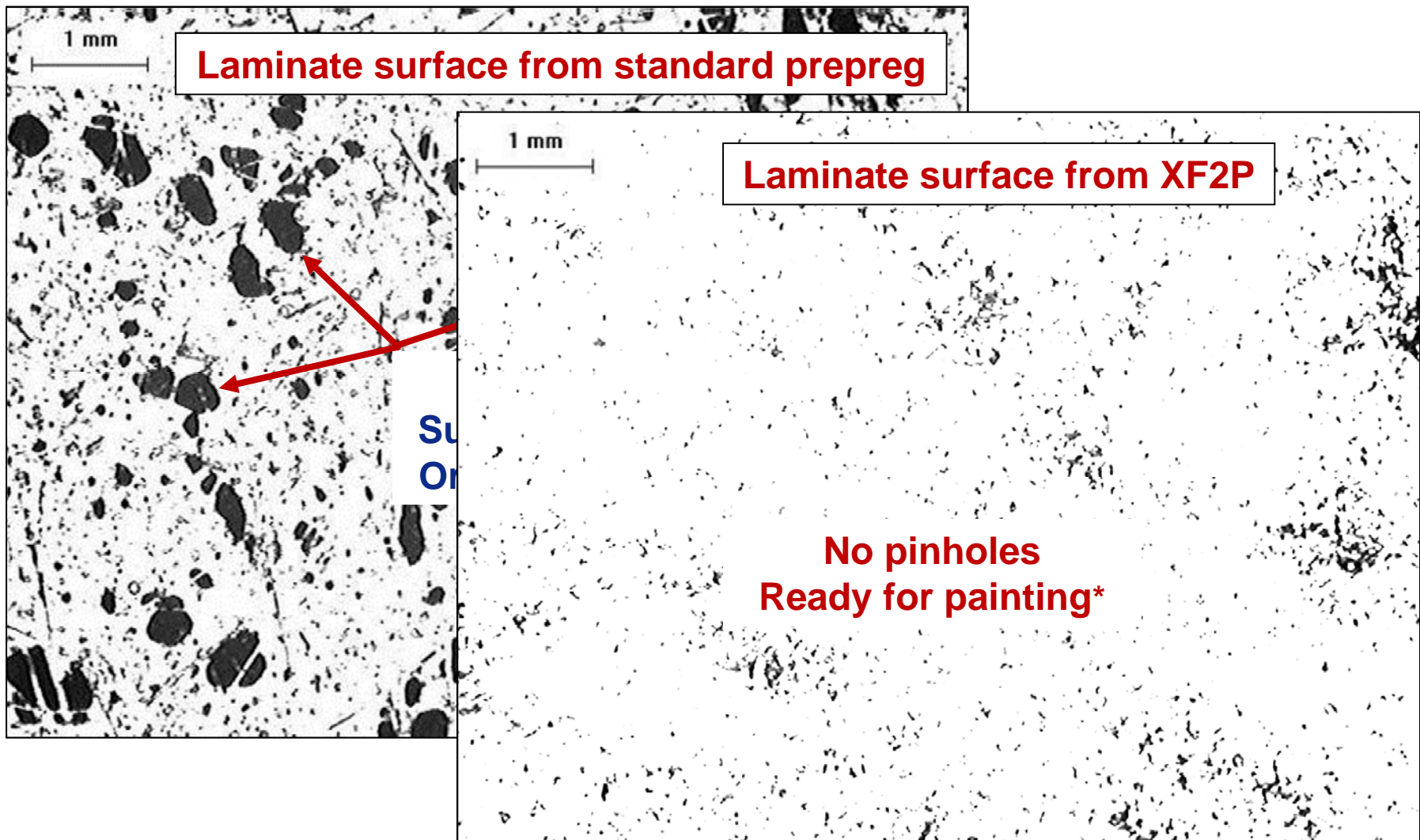
- Gel coats may be used to provide a good paint-ready surface
- Polyurethane paints may be used for the final surface

This process can be simplified by using specific shell prepregs

e.g. Products such as XF2P

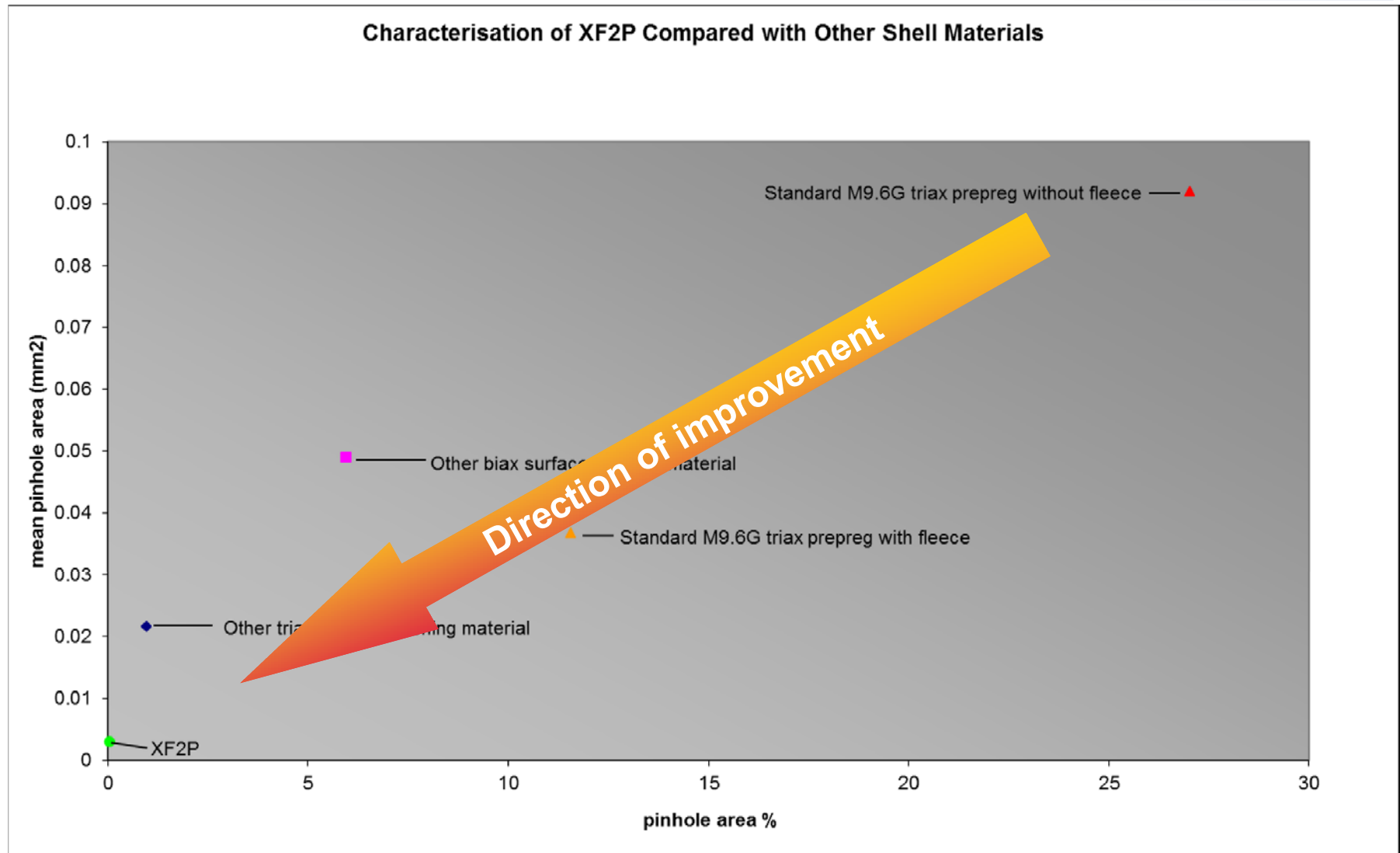
- Build the aerodynamic shell surface
- Eliminate the gel coat

XF2P: Surface Characterisation

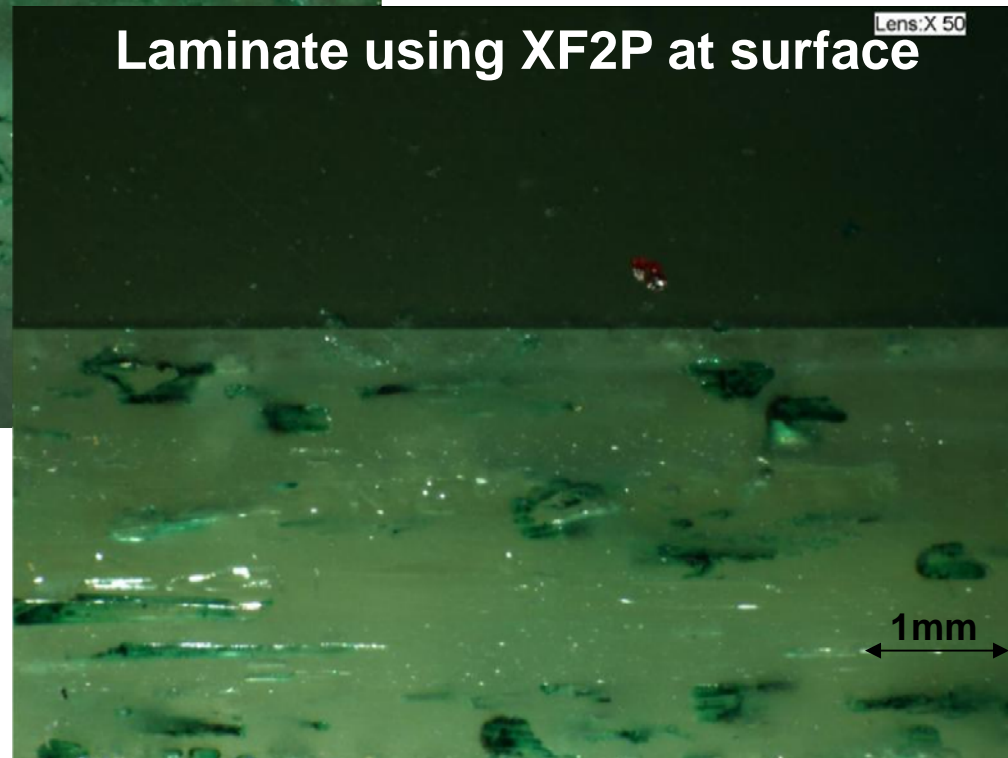
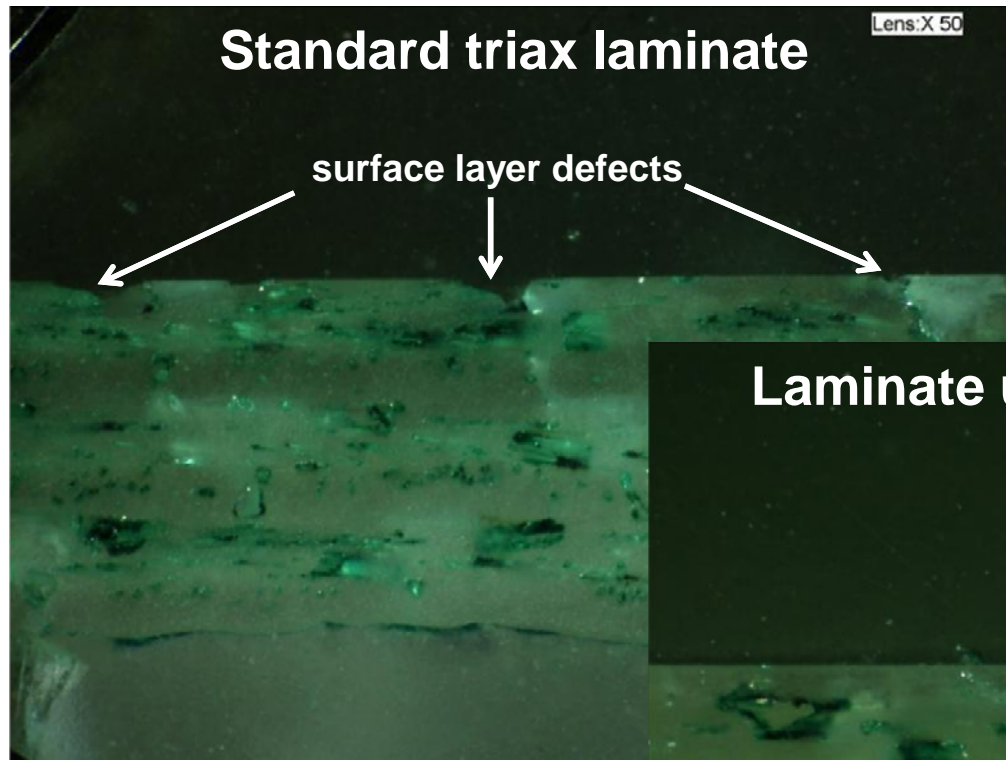


* After removal of release agent

Surface Porosity from Shell Materials

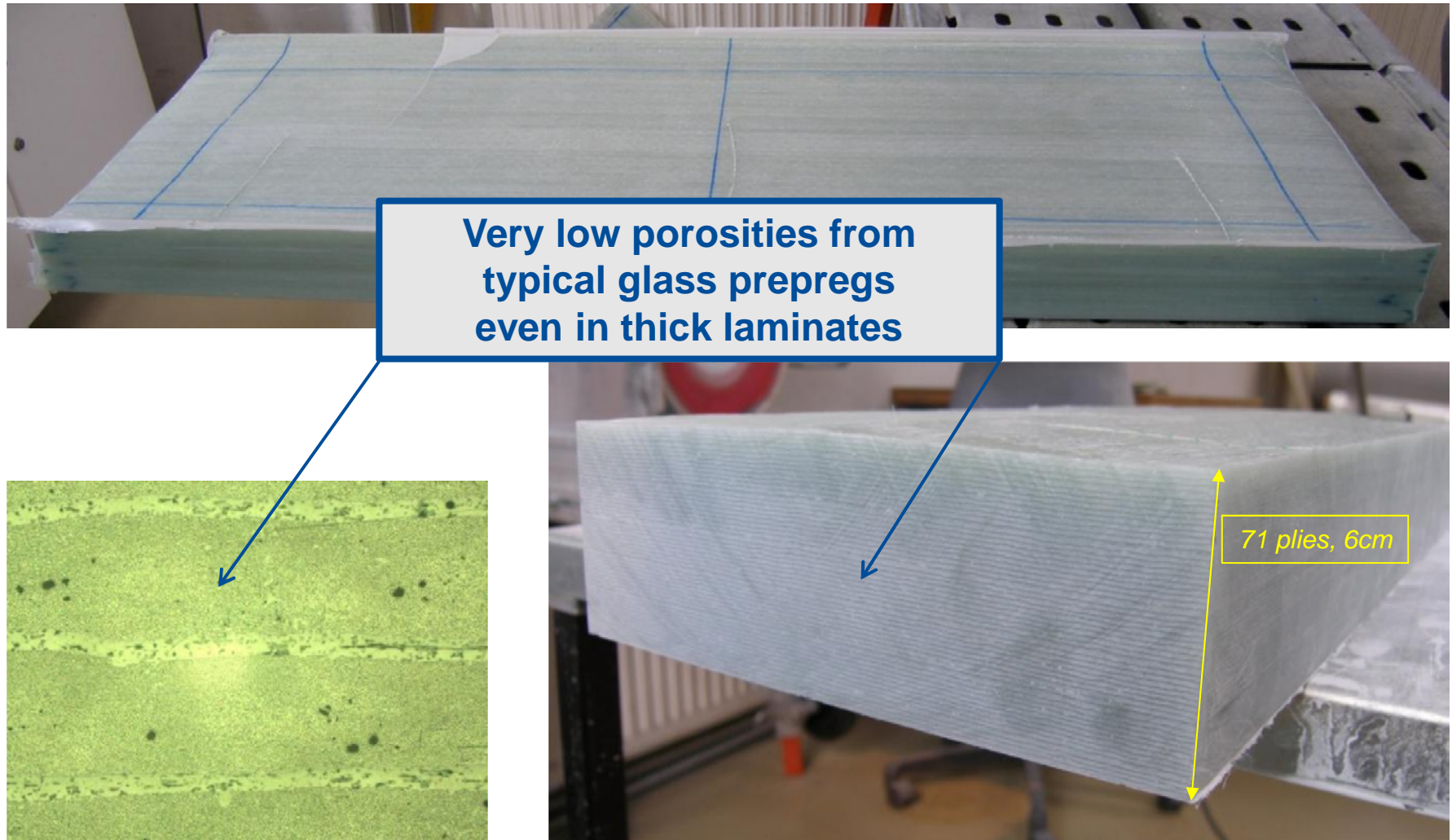


XF2P: Cross-sectional Analysis

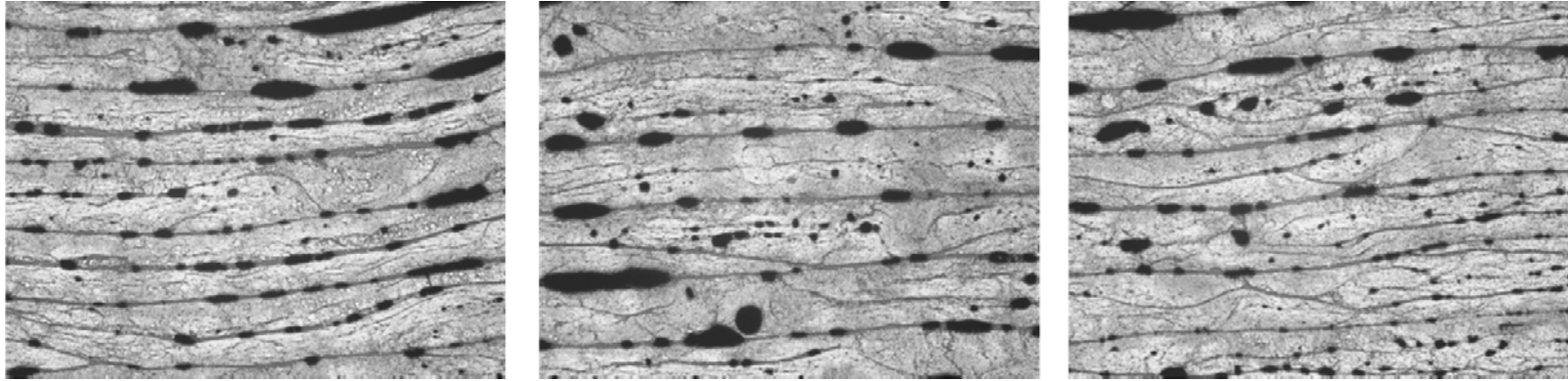


Prepregs for the Load-carrying Structure

Thick Glass Laminates using Prepregs



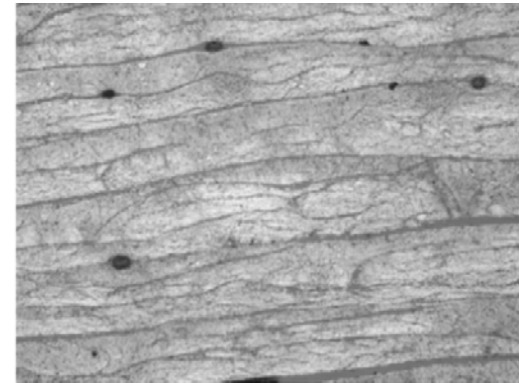
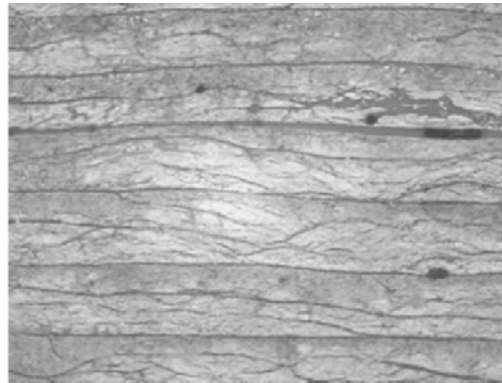
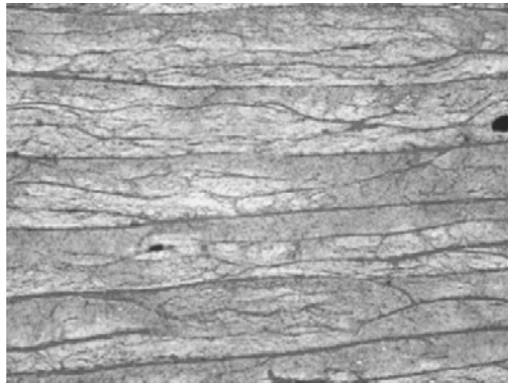
Thick Carbon Laminates – Conventional Technology



64 ply laminates using 600gsm carbon (HS)
prepreg and conventional technology
Porosity ~7%

Conventional prepregs are not optimised for thick carbon laminates

Thick Carbon Laminates – Optimised Architecture



Prepreg architecture designed for thick laminates
using **Hexcel patented technology**
Porosity $\ll 1\%$

**Optimised architecture in carbon UD prepregs
consistently minimises low porosity**

Prepreg and Infusion Mechanical Properties

Glass

Glass: Materials

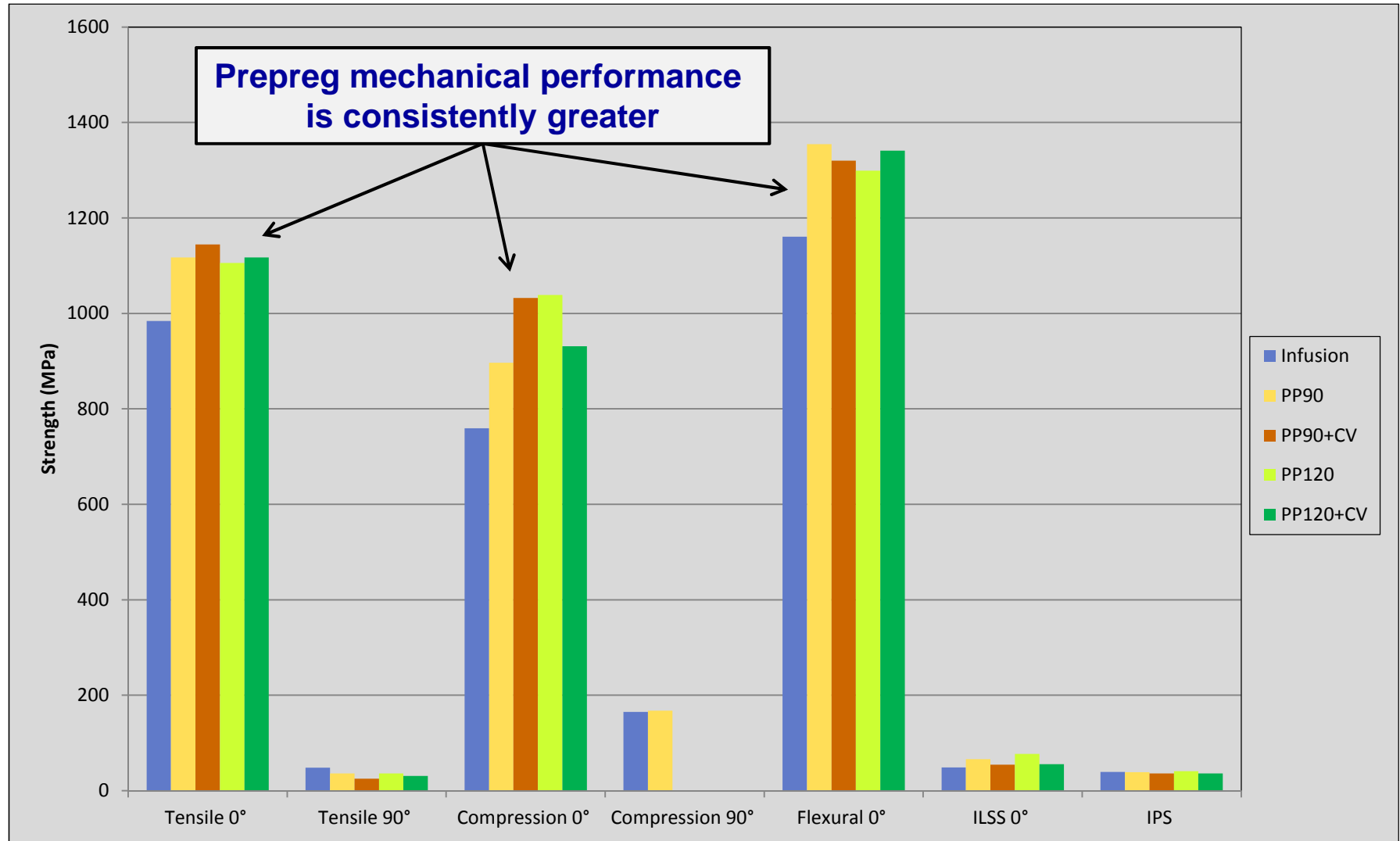
Infusion

- Reinforcement: LT1218 (UD1200 + slight reinforcement in 90°)
- Resin: Epikote RIM 135
- Cure at 90°C

Prepreg

- M9.6GLT/32%/1200(+CV)/G
- Cure at 90°C ('PP90') and 120°C ('PP120')

Glass: Mechanical Properties



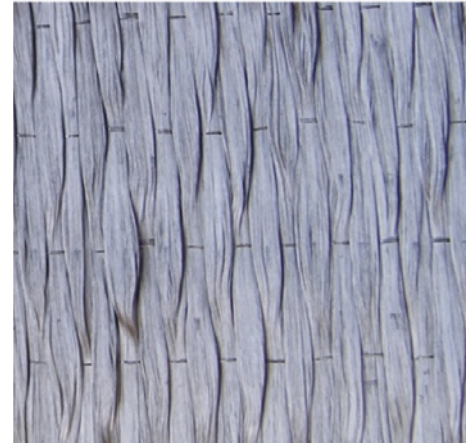
Prepreg and Infusion Mechanical Properties

Carbon

Carbon: Materials

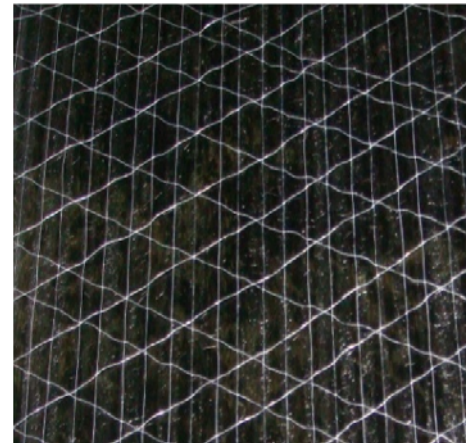
Infusion

- Reinforcement: UD600 low crimp T620
- Resin: Epikote RIM 135
- Cure at 90°C

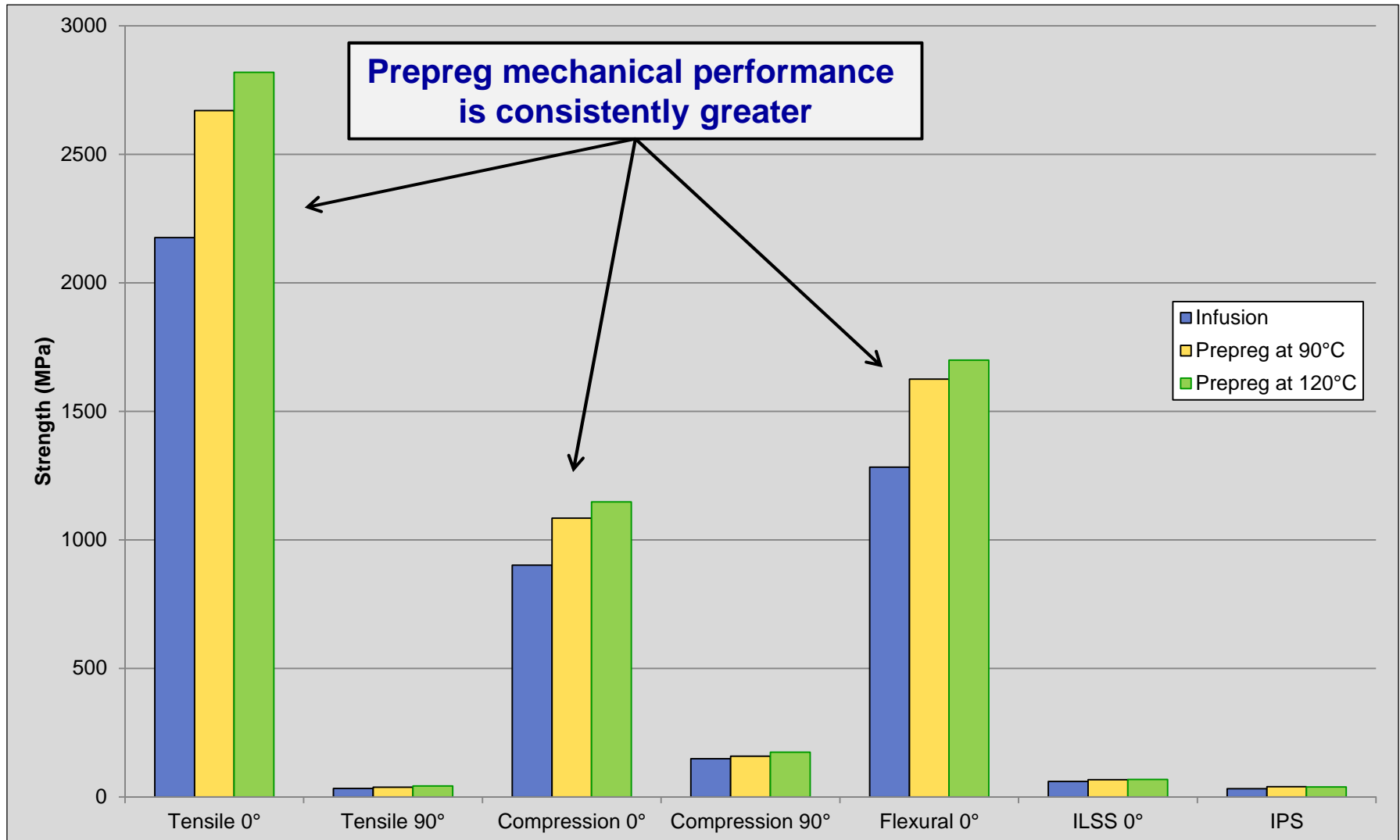


Prepreg

- M9.6GLT/35%/UD600+2P/T620+PES
- Cure at 90°C and 120°C



Carbon: Mechanical Properties

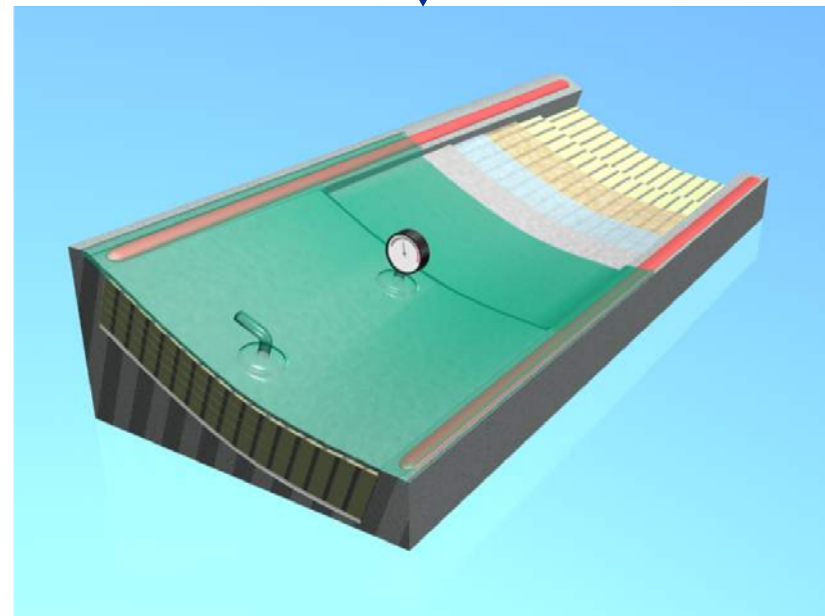
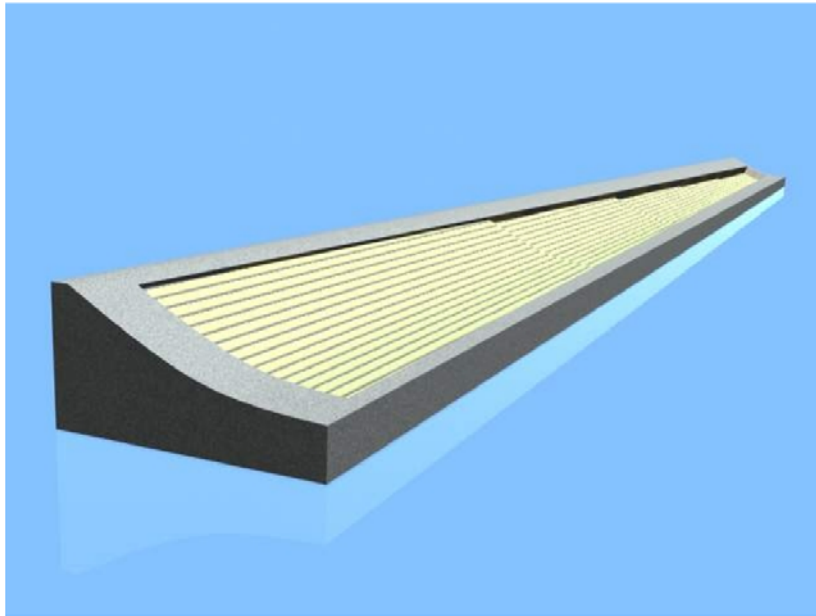


Use of Prepregs for the Construction of Spar Caps

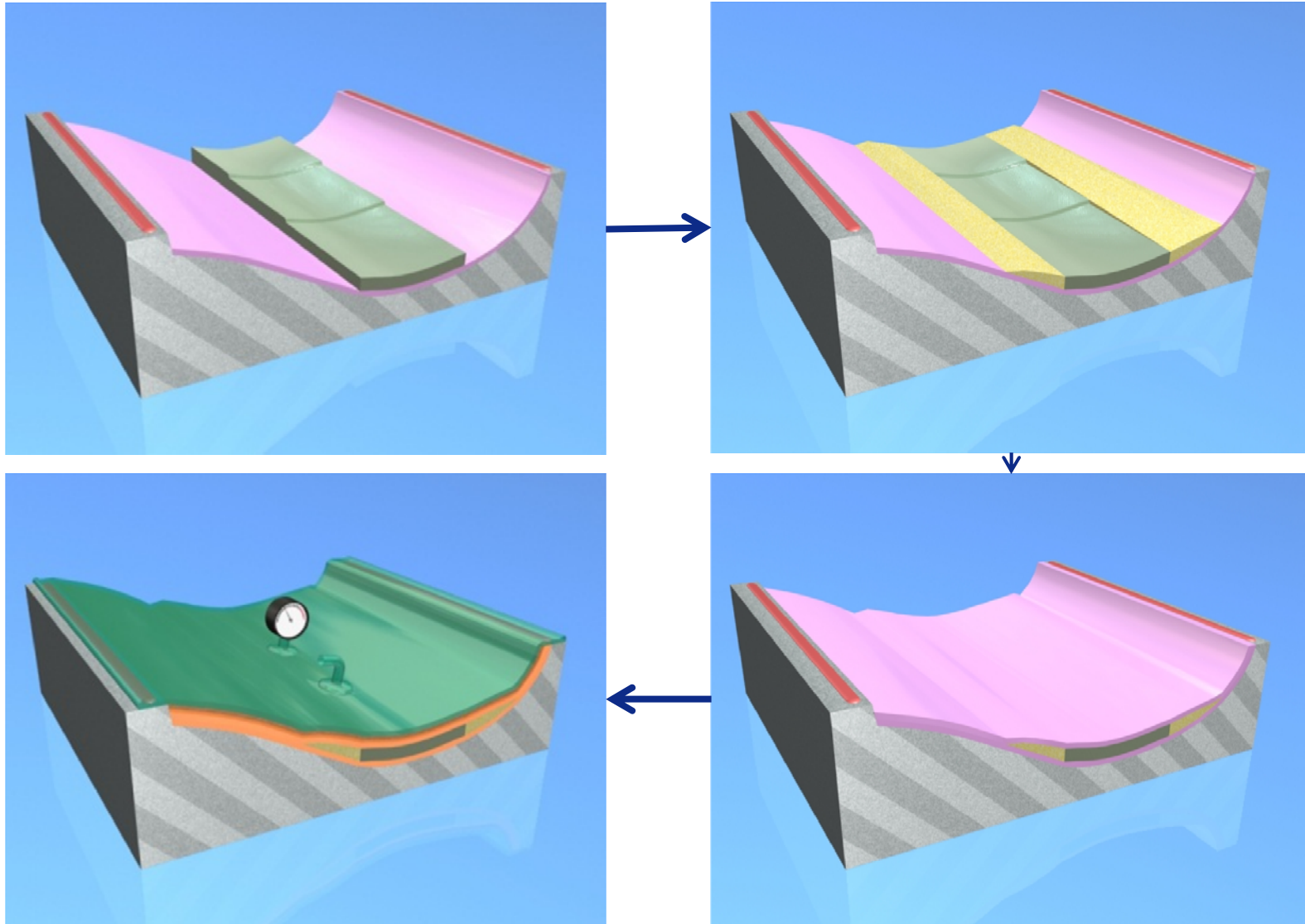
Example 1

**Pre-cured Spar Cap for Infusion
and Final Cure**

Spar Caps: Prepreg Layup and Cure

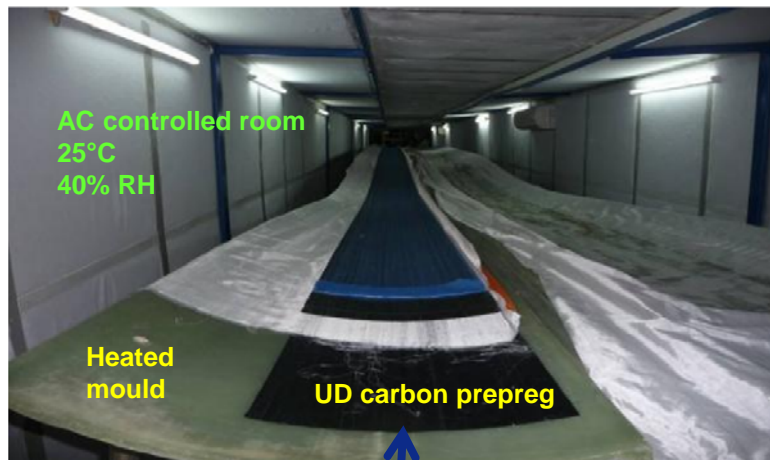


Spar Caps: Inclusion in an Infused Shell

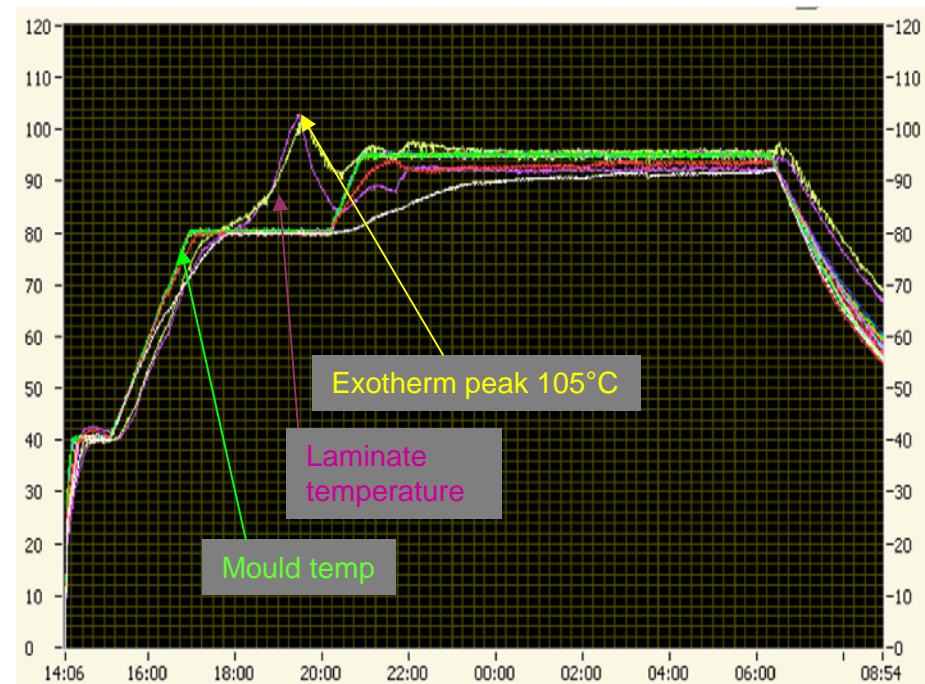


Case Study: Carbon Spar Cap at Half Scale

Carbon spar cap: length	25 m
width	0,40 m
thickness	22 mm
Number of plies	43
Material	M9.6/32%/500+8P/C

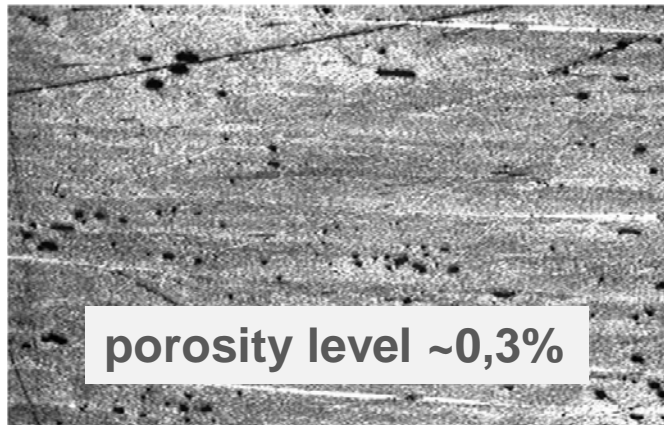


UD prepregs are ideally suited to automated layup



Case Study: Results

Spar cap Tg	106°C	
Average porosity	0,24%	
Highest porosity value	0,8%	(1/135 points)
Lowest porosity value	0%	(19/135 points)
Resin content	30%	



**Typical cross section of
cured laminate**



**Good adhesion of infused glass
on prepreg carbon laminate**

Use of Prepregs for the Construction of Spar Caps

Example 2

**Prepreg Spar Cap Co-infused in the Shell
with Final Co-cure**

Co-infusion: an Introduction

Co-infusion

The use of prepreg and infusion technologies in the same laminate with co-cure

Typical configuration

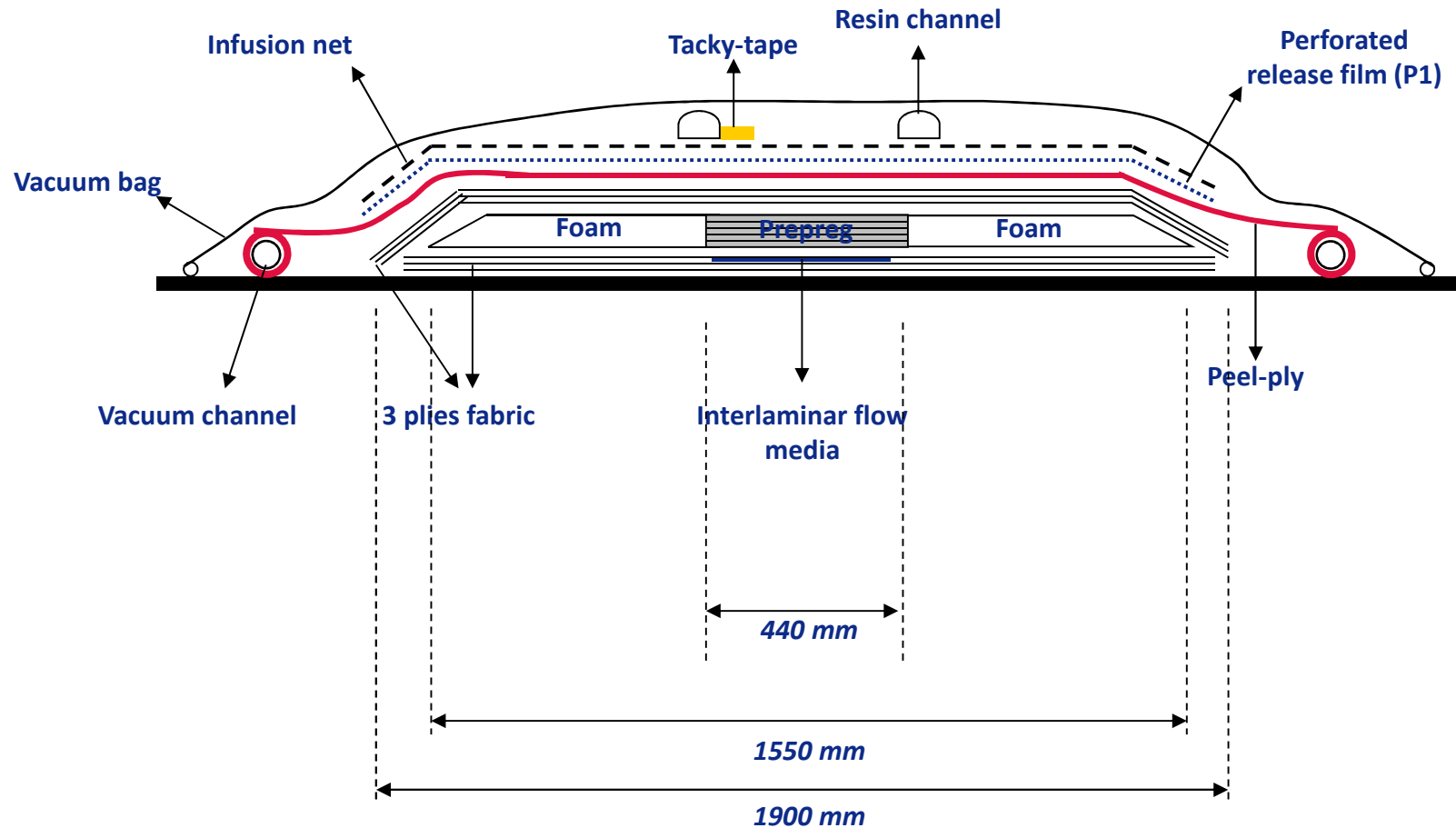
UD prepreg for the heavy load-carrying structure

Infusion of dry reinforcement for the remainder of the structure

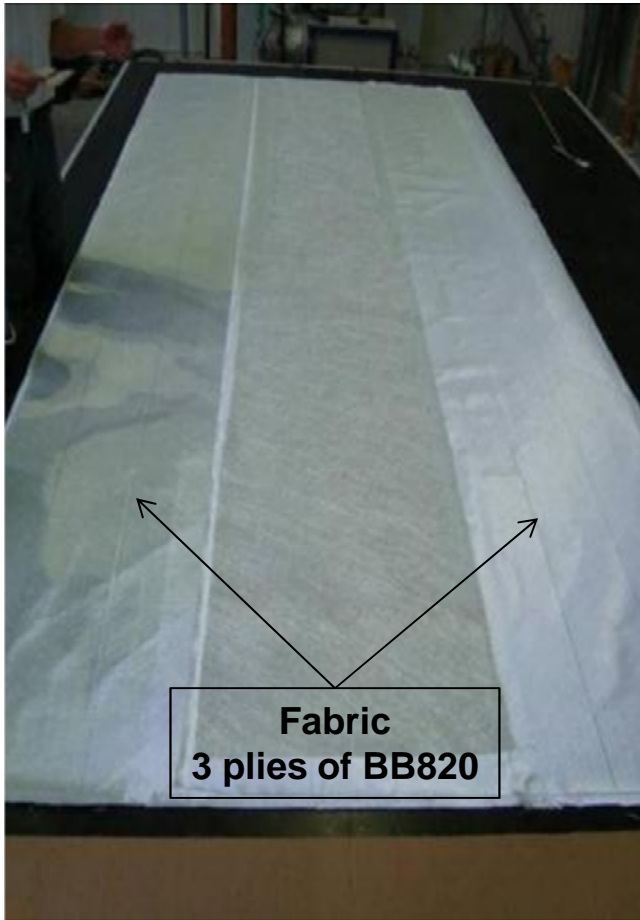
Cure of the whole assembly at the same time and temperature

Co-infusion: Case Study, Construction

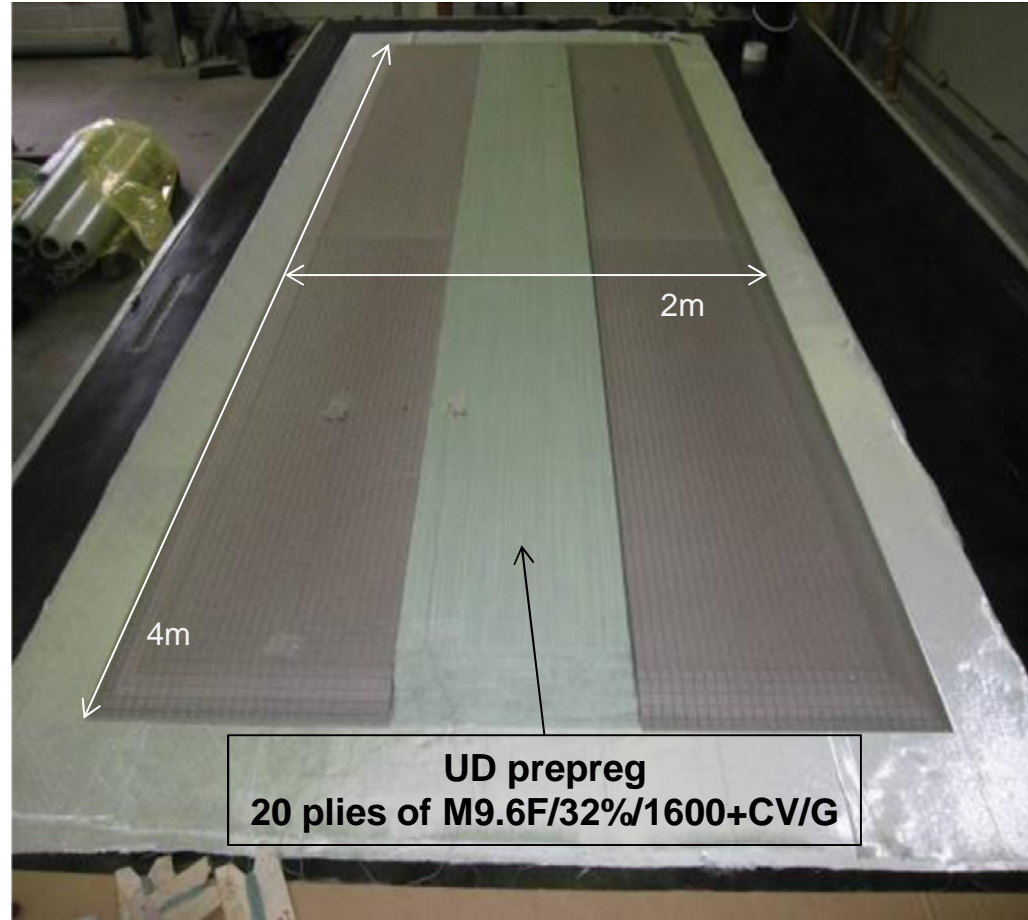
Demonstration on a 4 x 2m scale
UD prepreg with biax dry fabrics



Co-infusion: Case Study, Layup

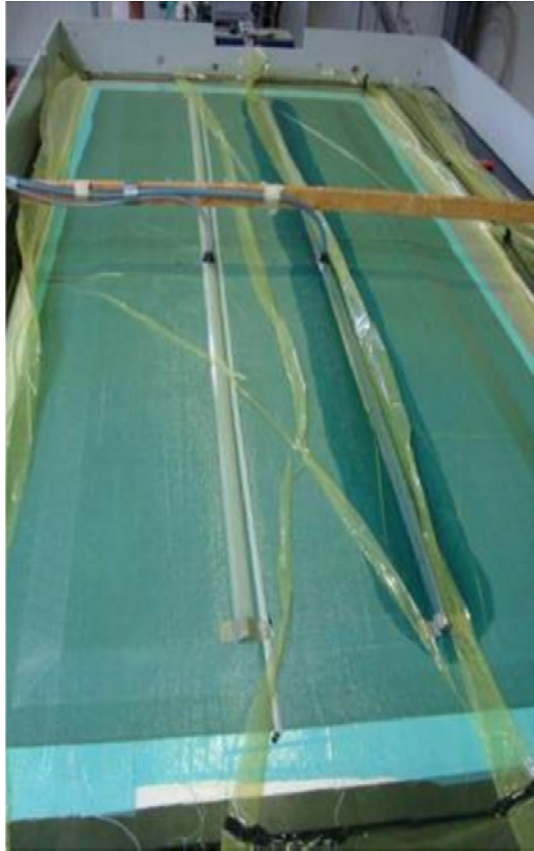


**Dry
reinforcements**

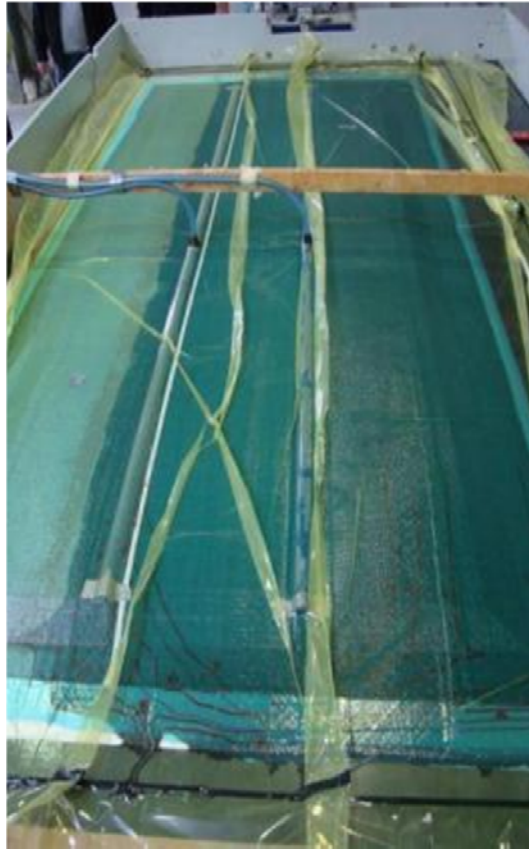


**Foam and UD prepreg
layers**

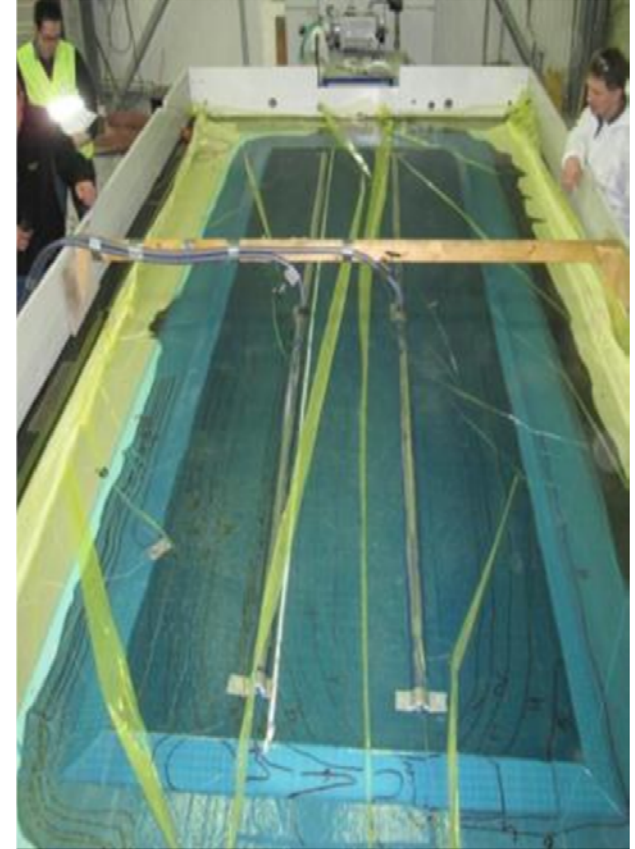
Co-infusion: Case Study, Infusion Process



1 min



12 min



22 min

Infusion time: ~25 min

Resin consumption: ~34 kg, Epikote RIM 135

Co-infusion: Case Study, Demoulding

Full impregnation of the laminate



Low porosity, high Tg

FV (%)		50
Porosity (%)	Side	0,7
	Middle	1,5
Tg (°C)	Top	75
	Middle	120
	Bottom	75
Cure cycle		6hrs 90°C

Co-infusion simplifies the production process, combining the best features of prepreg and infusion materials

Conclusions

Prepregs can be tailored for optimal wind blade manufacture

- **Lowest exotherm from matrix selection**
- **Shortest cure cycle from fast cure and low exotherm**
- **Minimal porosity from well-designed architecture**
 - **Both within the laminate and on the surface**
- **Reliable and full impregnation, even of carbon**

Prepregs are ideally suited for thick structural sections

Co-infusion simplifies the manufacturing process

- **It eliminates the separate steps for spar cap production**

Carbon and glass prepregs are ideally suited to heavy load-critical structures in wind blades

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